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Compressors	

Concerted experimental and numerical studies on axial flow fan rotor aerodynamics

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SYNOPSIS

The paper reports a concerted application of experimental (Laser Doppler Anemometry, LDA) and Computational Fluid Dynamics (CFD) techniques for an investigation of the high average total head rise axial flow fan rotor of non-free vortex design. The LDA data have been used as inlet boundary conditions for CFD calculations as well as a basis for verification of computational results. From a comparison of outlet LDA and CFD data, possible improvements in the computational technique have been pointed out. The interblade CFD results supplied guidelines for improvement in the rotor cascade design method.

NOTATIONS

Latin letters

```
C
          absolute velocity
          turbulent dissipation rate production constants
C_{EI}, C_{E2}
d
          prescribed Dirichelet boundary conditions
          residual forces vector
k
          turbulent kinetic energy
E
          blade chord
          normal vector Cartesian components
p
          radius
R = r/r, dimensionless radius
          blade pitch
          reference velocity (r \cdot \omega)
          relative velocity Cartesian components
          Reynolds stress tensor
           prescribed Neumann boundary conditions
Greek letters
          blade stagger angle
```

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