

DANIEL P. RAYMER

**Aircraft Design:
A Conceptual Approach**

AIAA
Education Series

J.S. PRZEMIENIECKI / SERIES EDITOR-IN-CHIEF

Aircraft Design: A Conceptual Approach

Daniel P. Raymer

President, Conceptual Research Corporation
Sylmar, California



EDUCATION SERIES

J. S. Przemieniecki

Series Editor-in-Chief

Air Force Institute of Technology

Wright-Patterson Air Force Base, Ohio

Published by

American Institute of Aeronautics and Astronautics, Inc.
370 L'Enfant Promenade, S.W., Washington, D.C. 20024

American Institute of Aeronautics and Astronautics, Inc., Washington, DC

Library of Congress Cataloging-in-Publication Data

Raymer, Daniel P.

Aircraft design:a conceptual approach/Daniel P. Raymer.

p. cm.—(AIAA education series)

Bibliography: p.

Includes index.

1. Airplanes—Design and construction. I. American Institute of Aeronautics and Astronautics. II. Title. III. Series.

TL671.2.R29 1989 629.134'1—dc20 89-14912 CIP

ISBN 0-930403-51-7

Second Edition, Sixth Printing

Copyright © 1992 by Daniel P. Raymer. Printed in the United States of America. No part of this publication may be reproduced, distributed, or transmitted in any form or by any means, or stored in a data base or retrieval system, without prior written permission of the publisher.

DISCLAIMER: The Author and the AIAA do not guarantee the accuracy of the information provided in this book, and it should not be referenced as an authoritative source for aircraft design data or methods.

DEDICATION

This book is dedicated to all who taught me, especially Lester Hendrix, Richard Hibma, Louis Hecq, Harry Scott, Richard Child, George Owl, Robert Maier, Ed McGachan, Doug Robinson, Steve White, Harvey Hoge, Michael Robinson, George Palmer, Henry Yang, Robert Swaim, C. T. Sun, Dave Schmidt, Bruce Reese, William Heiser, and Gordon Raymer (test pilot, aeronautical engineer and my father).

Thanks also to Rockwell North American Aircraft Operations for permission to use various illustrations. All other artwork is original, in the public domain, or copyrighted by AIAA.

Texts Published in the AIAA Education Series

- Re-Entry Vehicle Dynamics
Frank J. Regan, 1984
- Aerothermodynamics of Gas Turbine and Rocket Propulsion
Gordon C. Oates, 1984
- Aerothermodynamics of Aircraft Engine Components
Gordon C. Oates, Editor, 1985
- Fundamentals of Aircraft Combat Survivability Analysis and Design
Robert E. Ball, 1985
- Intake Aerodynamics
J. Seddon and E. L. Goldsmith, 1985
- Composite Materials for Aircraft Structures
Brian C. Hoskins and Alan A. Baker, Editors, 1986
- Gasdynamics: Theory and Applications
George Emanuel, 1986
- Aircraft Engine Design
Jack D. Mattingly, William Heiser, and Daniel H. Daley, 1987
- An Introduction to the Mathematics and Methods of Astrodynamics
Richard H. Battin, 1987
- Radar Electronic Warfare
August Golden Jr., 1988
- Advanced Classical Thermodynamics
George Emanuel, 1988
- Aerothermodynamics of Gas Turbine and Rocket Propulsion,
Revised and Enlarged
Gordon C. Oates, 1988
- Re-Entry Aerodynamics
Wilbur L. Hankey, 1988
- Mechanical Reliability: Theory, Models and Applications
B. S. Dhillon, 1988
- Aircraft Landing Gear Design: Principles and Practices
Norman S. Currey, 1988
- Gust Loads on Aircraft: Concepts and Applications
Frederic M. Hoblit, 1988
- Aircraft Design: A Conceptual Approach
Daniel P. Raymer, 1989
- Boundary Layers
A. D. Young, 1989
- Aircraft Propulsion Systems Technology and Design
Gordon C. Oates, Editor, 1989

Basic Helicopter Aerodynamics
J. Seddon, 1990

Introduction to Mathematical Methods in Defense Analyses
J. S. Przemieniecki, 1990

Space Vehicle Design
Michael D. Griffin and James R. French, 1991

Inlets for Supersonic Missiles
John J. Mahoney, 1991

Defense Analyses Software
J. S. Przemieniecki, 1991

Critical Technologies for National Defense
Air Force Institute of Technology, 1991

Orbital Mechanics
Vladimir A. Chobotov, 1991

Nonlinear Analysis of Shell Structures
Anthony N. Palazotto and Scott T. Dennis, 1992

Optimization of Observation and Control Processes
Veniamin V. Malyshev, Mikhail N. Krasilshikov, and Valeri I. Karlov,
1992

Aircraft Design: A Conceptual Approach
Second Edition
Daniel P. Raymer, 1992

Published by
American Institute of Aeronautics and Astronautics, Inc., Washington, DC

FOREWORD

As one of its major objectives, the AIAA Education Series is creating a comprehensive library of the established practices in aerospace design. *Aircraft Design: A Conceptual Approach* by Daniel P. Raymer provides an authoritative exposition of aircraft conceptual design. The great demand for the first edition of this new authoritative text on aircraft design has prompted the author to update and enlarge the text content into a second edition. In particular, Chapters 8 (Special Considerations in Configuration Layout), 13 (Propulsion), 17 (Performance and Flight Mechanics), and 21 (Conceptual Design Examples) have been extensively enlarged to cover some of the latest developments. The author's extensive experience with several aircraft companies supports the broad cross section of different views and approaches discussed in this comprehensive volume.

This textbook offers aircraft designers, design managers, and design instructors an industry perspective on the new aircraft concept development process, which basically consists of two major activities: design layout and design analysis. The whole process is described in a very comprehensive manner, tailored to serve as a college design textbook. However, only an elementary knowledge of mathematics is required to make full use of the text, for the book focuses on industry design practice rather than theoretical definitions. A simplified but complete set of first-order analytical methods is presented. The text covers every phase of conceptual design: configuration layout, payload considerations, aerodynamics, propulsion, structure and loads, weights, stability and control, handling qualities, performance, cost analysis, tradeoff analysis, and many other topics.

This latest text in the AIAA Education Series offers students, teachers, and practicing designers a unique source of information on current design practice in the U.S. aircraft industry—its science and art. To write a textbook on aircraft design is indeed a formidable task. Raymer has succeeded in creating a balanced text in which all the necessary topics needed to understand the design process are clearly described.

For many years *Aircraft Design: A Conceptual Approach* will be a valuable textbook for all who struggle with the fundamentals and intricacies of aircraft design.

J. S. PRZEMIENIECKI

Editor-in-Chief

AIAA Education Series

AUTHOR'S NOTE

There are two equally important aspects of aircraft design: design layout and design analysis. These very different activities attract different types of people. Some people love playing with numbers and computers, while others can't stop doodling on every piece of paper within reach.

This book was written to fill a perceived need for a textbook in which both aircraft analysis and design layout are covered equally, and the interactions between these two aspects of design are explored in a manner consistent with industry practice.

This book is not intended to be definitive on the subject of aircraft analysis. The analysis techniques presented are simplified to permit the student to experience the whole design process in a single course, including the key concepts of trade studies and aircraft optimization.

No textbook can contain the methods actually used in industry, which tend to be proprietary and highly computerized. When the student goes into an industry or government design job, the more sophisticated methods of his or her chosen specialty will be better understood in the broader context of the whole of design as presented here.

One key area in which this book differs from prior aircraft design books is in the chapters on aircraft configuration layout. The actual development of the aircraft design drawing is not a trivial task of drafting based upon the analysis results, but rather is a key element of the overall design process and ultimately determines the performance, weight, and cost of the aircraft.

The ability to visualize and draw a new aircraft that has a streamlined aerodynamic shape, an efficient internal layout, yet satisfies an incredible number of real-world constraints and design specifications is a rare talent that takes years to cultivate. While to some extent good designers are "born, not made," a number of concepts and techniques in aircraft configuration layout can be taught, and are covered here.

Writing this book has been an educating and humbling experience. It is my sincere wish that it help aspiring aircraft designers to "learn the ropes" more quickly.

This second edition of *AIRCRAFT DESIGN: A Conceptual Approach* offers several new subjects, including production methods, post-stall maneuver, an update on VSTOL, and a brief introduction to engine cycle analysis. Also, typographical and technical errors from the first edition are corrected.

A key difference in the second edition is Chapter 21, the Conceptual Design Examples. These are reworked to better serve as examples for the chapters of the book. The second example illustrates the use of RDS, a PC-based design, sizing and performance program now available from AIAA. RDS uses the methods in this book, and permits rapid design, analysis, and trade studies.

AIAA and the author would like to thank the many people who have offered constructive suggestions for this second edition, as well as the more than 7000 students and working engineers who made the first edition an AIAA best seller.



Display model of an Advanced Supercruise Fighter Concept (Ref. 13). Photo courtesy of Rockwell International North American Aircraft Operations.

知
飛
機

TABLE OF CONTENTS

Author's Note	xiii
Chapter 1. Design—A Separate Discipline	
1.1 What Is Design?	1
1.2 Introduction to the Book	1
Chapter 2. Overview of the Design Process	
2.1 Introduction	3
2.2 Phases of Aircraft Design	4
2.3 Aircraft Conceptual Design Process	7
Chapter 3. Sizing from a Conceptual Sketch	
3.1 Introduction	11
3.2 Takeoff-Weight Buildup	11
3.3 Empty-Weight Estimation	12
3.4 Fuel-Fraction Estimation	14
3.5 Takeoff-Weight Calculation	23
3.6 Design Example: ASW Aircraft	24
Chapter 4. Airfoil and Geometry Selection	
4.1 Introduction	33
4.2 Airfoil Selection	33
4.3 Wing Geometry	47
4.4 Biplane Wings	65
4.5 Tail Geometry and Arrangement	67
Chapter 5. Thrust-to-Weight Ratio and Wing Loading	
5.1 Introduction	77
5.2 Thrust-to-Weight Ratio	78
5.3 Wing Loading	84
5.4 Selection of Thrust-to-Weight and Wing Loading	99
Chapter 6. Initial Sizing	
6.1 Introduction	101
6.2 Rubber-Engine Sizing	102
6.3 Fixed-Engine Sizing	108
6.4 Geometry Sizing	109
6.5 Control-Surface Sizing	113
Chapter 7. Configuration Layout and Loft	
7.1 Introduction	117
7.2 End Products of Configuration Layout	117
7.3 Conic Lofting	123
7.4 Conic Fuselage Development	129
7.5 Flat-Wrap Fuselage Lofting	135
7.6 Circle-to-Square Adapter	136
7.7 Fuselage Loft Verification	137
7.8 Wing/Tail Layout and Loft	139
7.9 Aircraft Layout Procedures	149
7.10 Wetted Area Determination	150
7.11 Volume Determination	152

Chapter 8. Special Considerations in Configuration Layout	
8.1 Introduction	155
8.2 Aerodynamic Considerations	155
8.3 Structural Considerations	158
8.4 Radar Detectability	165
8.5 Infrared Detectability	170
8.6 Visual Detectability	171
8.7 Aural Signature	171
8.8 Vulnerability Considerations	172
8.9 Crashworthiness Considerations	174
8.10 Producibility Considerations	175
8.11 Maintainability Considerations	179
Chapter 9. Crew Station, Passengers, and Payload	
9.1 Introduction	181
9.2 Crew Station	181
9.3 Passenger Compartment	185
9.4 Cargo Provisions	186
9.5 Weapons Carriage	188
9.6 Gun Installation	191
Chapter 10. Propulsion and Fuel System Integration	
10.1 Introduction	193
10.2 Propulsion Selection	193
10.3 Jet-Engine Integration	196
10.4 Propeller-Engine Integration	220
10.5 Fuel System	226
Chapter 11. Landing Gear and Subsystems	
11.1 Introduction	229
11.2 Landing Gear Arrangements	229
11.3 Tire Sizing	233
11.4 Shock Absorbers	239
11.5 Castoring-Wheel Geometry	246
11.6 Gear-Retraction Geometry	247
11.7 Seaplanes	250
11.8 Subsystems	252
Chapter 12. Aerodynamics	
12.1 Introduction	257
12.2 Aerodynamic Forces	258
12.3 Aerodynamic Coefficients	262
12.4 Lift	263
12.5 Parasite (Zero-Lift) Drag	280
12.6 Drag Due to Lift (Induced Drag)	297
12.7 Aerodynamic Codes and Computational Fluid Dynamics (CFD)	305
Chapter 13. Propulsion	
13.1 Introduction	313
13.2 Jet-Engine Thrust Considerations	315
13.3 Turbojet Installed Thrust	317
13.4 Thrust-Drag Bookkeeping	317
13.5 Installed-Thrust Methodology	318
13.6 Piston-Engine Performance	325
13.7 Turboprop Performance	331

Chapter 14. Structures and Loads	
14.1	Introduction 333
14.2	Loads Categories 334
14.3	Air Loads 335
14.4	Inertial Loads 347
14.5	Power-Plant Loads 348
14.6	Landing-Gear Loads 348
14.7	Structures Fundamentals 349
14.8	Material Selection 354
14.9	Material Properties 357
14.10	Structural-Analysis Fundamentals 369
14.11	Finite-Element Structural Analysis 389
Chapter 15. Weights	
15.1	Introduction 395
15.2	Approximate Group Weights Method 399
15.3	Statistical Group Weights Method 399
15.4	Additional Considerations in Weights Estimation 407
Chapter 16. Stability, Control, and Handling Qualities	
16.1	Introduction 411
16.2	Coordinate Systems and Definitions 413
16.3	Longitudinal Static Stability and Control 414
16.4	Lateral-Directional Static Stability and Control 433
16.5	Stick-Free Stability 441
16.6	Effects of Flexibility 442
16.7	Dynamic Stability 443
16.8	Quasi-Steady State 446
16.9	Inertia Coupling 448
16.10	Handling Qualities 449
Chapter 17. Performance and Flight Mechanics	
17.1	Introduction and Equations of Motion 455
17.2	Steady Level Flight 457
17.3	Steady Climbing and Descending Flight 463
17.4	Level Turning Flight 467
17.5	Gliding Flight 471
17.6	Energy-Maneuverability Methods 475
17.7	Operating Envelope 483
17.8	Takeoff Analysis 486
17.9	Landing Analysis 489
17.10	Other Fighter Performance Measures of Merit 491
Chapter 18. Cost Analysis	
18.1	Introduction 501
18.2	Elements of Life-Cycle Cost 503
18.3	Cost-Estimating Methods 505
18.4	RDT&E and Production Costs 506
18.5	Operations and Maintenance Costs 510
18.6	Cost Measures of Merit (Military) 514
18.7	Airline Economics 514

Chapter 19. Sizing and Trade Studies	
19.1 Introduction	519
19.2 Detailed Sizing Methods	519
19.3 Improved Conceptual Sizing Methods	520
19.4 Sizing Matrix and Carpet Plots	525
19.5 Trade Studies	532
Chapter 20. VTOL Aircraft Design	
20.1 Introduction	537
20.2 VTOL Terminology	538
20.3 Fundamental Problems of VTOL Design	538
20.4 VTOL Jet-Propulsion Options	541
20.5 Vectoring-Nozzle Types	547
20.6 Suckdown and Fountain Lift	551
20.7 Recirculation and Hot-Gas Ingestion	552
20.8 VTOL Footprint	553
20.9 VTOL Control	554
20.10 VTOL Propulsion Considerations	555
20.11 Weight Effects of VTOL	556
20.12 Sizing Effects of VTOL	557
Chapter 21. Conceptual Design Examples	
21.1 Introduction	559
21.2 Single-Seat Aerobatic	559
21.3 Lightweight Supercruise Fighter	603
Appendix A	
A.1 Conversion Tables	658
A.2 Standard Atmosphere and Shock Tables	660
A.3 Airfoil Data	687
A.4 Typical Engine Performance Curves	717
A.5 Design Requirements and Specifications	731
References	735
Subject Index	739

DESIGN—A SEPARATE DISCIPLINE

1.1 WHAT IS DESIGN?

Aircraft design is a separate discipline of aeronautical engineering—different from the analytical disciplines such as aerodynamics, structures, controls, and propulsion. An aircraft designer needs to be well versed in these and many other specialties, but will actually spend little time performing such analysis in all but the smallest companies. Instead, the designer's time is spent doing something called "design," creating the geometric description of a thing to be built.

To the uninitiated, "design" looks a lot like "drafting" (or in the modern world, "computer-aided drafting"). The designer's product is a drawing, and the designer spends the day hunched over a drafting table or computer terminal. However, the designer's real work is mostly mental.

If the designer is talented, there is a lot more than meets the eye on the drawing. A good aircraft design seems to miraculously glide through subsequent evaluations by specialists without major changes being required. Somehow, the landing gear fits, the fuel tanks are near the center of gravity, the structural members are simple and lightweight, the overall arrangement provides good aerodynamics, the engines install in a simple and clean fashion, and a host of similar detail seems to fall into place.

This is no accident, but rather the product of a lot of knowledge and hard work by the designer. This book was written primarily to provide the basic tools and concepts required to produce good designs which will survive detailed analysis with minimal changes.

Other key players participate in the design process. Design is not just the actual layout, but also the analytical processes used to determine what should be designed and how the design should be modified to better meet the requirements. In a small company, this may be done by the same individuals who do the layout design. In the larger companies, aircraft analysis is done by the sizing and performance specialists with the assistance of experts in aerodynamics, weights, propulsion, stability, and other technical specialties.

In this book, the design layout techniques are discussed primarily in Chapters 4–11, while the analysis and optimization methods are presented in Chapters 12–19.

1.2 INTRODUCTION TO THE BOOK

This book describes the process used to develop a credible aircraft conceptual design from a given set of requirements. As a part of the AIAA

Education Series, the book is written primarily for the college student. Every effort has been made to achieve a self-contained book.

In an aircraft company, the designer can ask a functional specialist for a reasonable initial tire size, inlet capture area, weight savings due to the use of composites, or similar estimates. Such specialists are not available at most universities. This book thus gives various "rule-of-thumb" approximations for initial estimation of design parameters.

The book has 21 chapters, and approximately follows the actual design sequence. Chapters 2 and 3 provide an overall introduction to the design process. Chapter 2 discusses how the conceptual design process works, and how it fits into the overall process of aircraft development. Chapter 3 presents a "first-pass" design procedure to familiarize the reader with the essential concepts of design, including design layout, analysis, takeoff-weight estimation, and trade studies.

In Chapters 4-11 the techniques for the development of the initial configuration layout are presented. These include the conceptual sketch, initial sizing, wing geometry selection, lofting, inboard layout, and integration of propulsion, crew station, payload/passenger compartment, fuel system, landing gear, and considerations for observability, producibility, and supportability. While the text implies that the design is done on a drafting board, it should be understood that in major aircraft companies today most aircraft design work is done on a computer-aided design system. However, the same basic design techniques are used whether on a drafting table or computer scope.

Chapters 12-19 address the analysis, sizing, and optimization of the design layout. Various chapters discuss aerodynamics, weights, installed propulsion characteristics, stability and control, performance, cost, and sizing. Optimization based upon design requirements is introduced in a section on trade studies.

These methods are simplified to allow rapid design analysis by students. No college textbook can contain the methods actually used by major aircraft companies, which tend towards highly sophisticated computer programs operated by specialists. Simplified analysis methods allow the student more time to experience the all-important optimization and iteration process.

Chapter 20 presents an overview of VTOL aircraft design. This material builds upon the methods for conventional aircraft design. However, VTOL introduces additional considerations that affect the design layout and analysis.

The last chapter, 21, contains two complete design project examples which use the methods presented in the previous chapters. These are provided instead of numerous example calculations throughout the text to illustrate how the different aspects of design fit together as a whole.

The appendices contain information useful in conceptual design, such as conversion tables, atmosphere and shock tables, and data on airfoils and engines. Also included is a summary of the current civil and military design requirements and specifications, which have been taken primarily from Federal Aviation Regulations (FAR) and Military Specifications (Mil-Specs).

OVERVIEW OF THE DESIGN PROCESS

2.1 INTRODUCTION

Those involved in design can never quite agree as to just where the design process begins. The designer thinks it starts with a new airplane concept. The sizing specialist knows that nothing can begin until an initial estimate of the weight is made. The customer, civilian or military, feels that the design begins with requirements.

They are all correct. Actually, design is an iterative effort, as shown in the "Design Wheel" of Fig. 2.1. Requirements are set by prior design trade studies. Concepts are developed to meet requirements. Design analysis frequently points toward new concepts and technologies, which can initiate a whole new design effort. However a particular design is begun, all of these activities are equally important in producing a good aircraft concept.

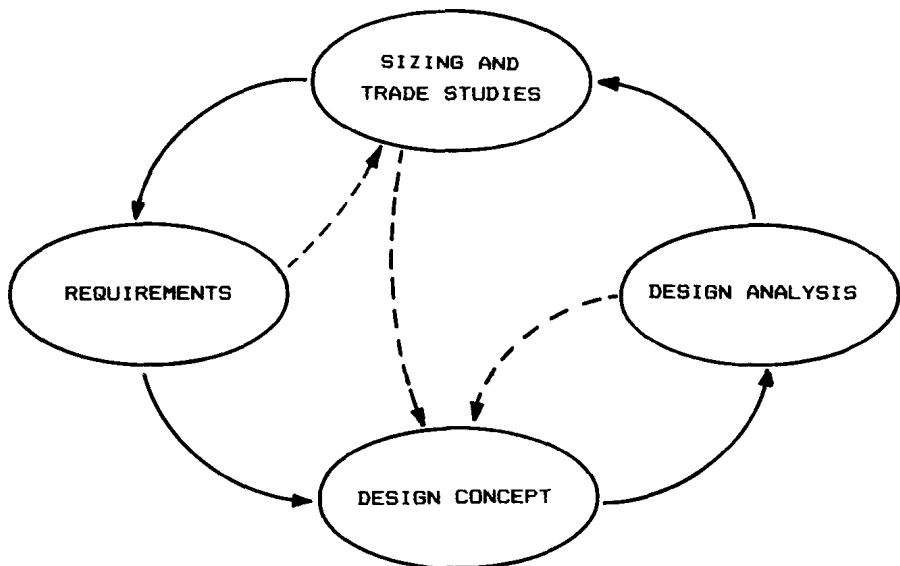


Fig. 2.1 The design wheel.

2.2 PHASES OF AIRCRAFT DESIGN

Conceptual Design

Aircraft design can be broken into three major phases, as depicted in Fig. 2.2. Conceptual design is the primary focus of this book. It is in conceptual design that the basic questions of configuration arrangement, size and weight, and performance are answered.

The first question is, "Can an affordable aircraft be built that meets the requirements?" If not, the customer may wish to relax the requirements.

Conceptual design is a very fluid process. New ideas and problems emerge as a design is investigated in ever-increasing detail. Each time the latest design is analyzed and sized, it must be redrawn to reflect the new gross weight, fuel weight, wing size, engine size, and other changes. Early wind-tunnel tests often reveal problems requiring some changes to the configuration. The steps of conceptual design are described later in more detail.

Preliminary Design

Preliminary design can be said to begin when the major changes are over. The big questions such as whether to use a canard or an aft tail have been resolved. The configuration arrangement can be expected to remain about as shown on current drawings, although minor revisions may occur. At some point late in preliminary design, even minor changes are stopped when a decision is made to freeze the configuration.

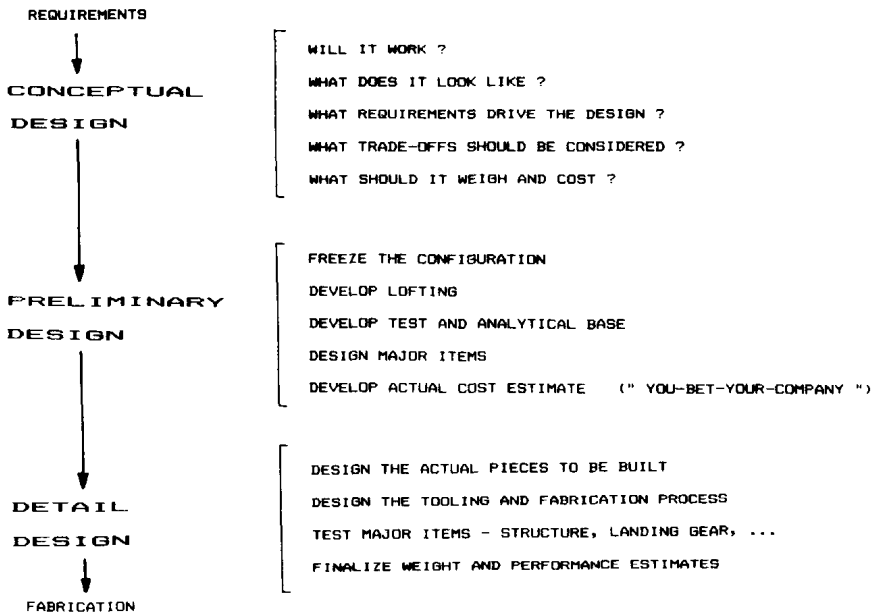


Fig. 2.2 Three phases of aircraft design.