Juvinall's

FUNDAMENTALS

OF MACHINE COMPONENT DESIGN

SI VERSION

ROBERT C. JUVINALL KURT M. MARSHEK

WILEY

JUVINALL'S FUNDAMENTALS OF MACHINE COMPONENT DESIGN

GLOBAL EDITION SI VERSION

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Appendix A-1a Conversion Factors for British Gravitational, English, and SI Units

	British Gravitational	,	
Quantity	and English Units ^a	SI Unit ^a	Conversion Factor Equalities
Length	inch (in. or ")	meter (m)	*1 in. = $0.0254 \text{ m} = 25.4 \text{ mm}$
	foot (ft or ')	meter (m)	*1 ft = $0.3048 \text{ m} = 304.8 \text{ mm}$
	mile (mi U.S. statute)	kilometer (km)	1 mile = $1.609 \text{ km} = 1609 \text{ m}$
Volume	gallon (gal U.S.)	meter ³ (m ³)	1 gal = $0.003785 \text{ m}^3 = 3.785 \text{ liters}$
Force (weight)	pound (lb)	newton ^d (N)	1 lb = 4.448 N
Torque	pound-foot (lb·ft)	newton-meter (N·m)	1 lb·ft = 1.356 N·m
Work, Energy	foot-pound (ft·lb)	joule ^c (J)	1 ft·lb = 1.356 J
Power	foot-pound/second (ft·lb/s)	watt ^g (W)	1 ft·lb/s = 1.356 W
	horsepower ^b (hp)	kilowatt (kW)	1 hp = 0.746 kW
Stress,	pounds/in. ² (psi)	pascale (Pa)	1 psi = 6895 Pa
Pressure	thousand pounds/in.2 (ksi)	megapascal (MPa)	1 ksi = 6.895 MPa
Mass (British)	slug ^f	kilogram (kg)	1 slug = 14.59 kg
Mass (English)	Ibm ^h	kilogram (kg)	1 lbm = $0.454 \text{ kg} = 454 \text{ grams}$

aThe larger unit is underlined. b1 hp = 550 ft · 1b/s; c1 J = 1 N·m; d1 N = 1 kg·m/s²; c1 Pa = 1 N/m²; f1 slug = 1 lb·s²/ft; g1 W = 1 J/s; h1 slug = 32.2 lbm.

*An exact definition.

Appendix C-1 Physical Properties of Common Metals

	Modulus of	lus of	Modu	Modulus of				Coeffic	Coefficient of				
	Elasticity, E	icity,	Rigi	Rigidity,		Unit	Density,		Thermal Expansion, α	Thermal Conductivity	nal tivity	Specific Heat	Heat
Metal	Mpsi	GPa	Mpsi	GPa	Ratio, v	(lb/in. ³)	$(\mathrm{Mg/m}^3)$	$10^{-6}/^{\circ}$ F	10 ₋ e/c	Btu/h-ft-°F	W/m-°C	Btu/lbm-°F	J/kg-°C
Aluminum alloy	10.4^{a}	72	3.9	27	0.32	0.10	2.8	12.0	22	100	173	0.22	920
Beryl. copper	18.5	127	7.2	50	0.29	0.30	8.3	9.3	17	85	147	0.10	420
Brass, Bronze	16	110	0.9	41	0.33	0.31	8.7	10.5	19	45	78	0.10	420
Copper	17.5	121	9.9	46	0.33	0.32	8.9	9.4	17	220	381	0.10	420
Iron, gray cast ^b	15	103	0.9	41	0.26	0.26	7.2	6.4	12	29	50	0.13	540
Magnesium alloy	6.5	45	2.4	17	0.35	0.065	1.8	14.5	26	55	95	0.28	1170
Nickel alloy	30	207	11.5	79	0.30	0.30	8.3	7.0	13	12	21	0.12	200
Steel, carbon	30	207	11.5	79	0.30	0.28	7.7	6.7	12	27	47	0.11	460
Steel, alloy	30	207	11.5	79	0.30	0.28	7.7	6.3	11	22	38	0.11	460
Steel, stainless	27.5	190	10.6	73	0.30	0.28	7.7	8.0	14	12	21	0.11	460
Titanium alloy	16.5	114	6.2	43	0.33	0.16	4.4	4.9	6	7	12	0.12	200
Zinc alloy	12	83	4.5	31	0.33	0.24	9.9	15.0	27	49	111	0.11	460

^aValues given are representative. Exact values may vary with composition and processing, sometimes greatly. ^bSee Appendix C-3 for more detailed elastic properties of cast irons.

Note: See Appendix C-18 for physical properties of some plastics.

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PREFACE

This book is intended as a text for first courses in Mechanical Engineering Design and as a reference for practicing engineers. It is assumed that the user has had basic courses in Mechanics, Strength of Materials, and Materials Properties. However, the first nine chapters of the book (Part I) serve to review as well as extend this basic background. The remaining chapters (Part II) deal with the application of these fundamentals to specific machine components.

Features of the Global Edition of the text include:

- *Homework problems*—A large number of new problems are added that help the student focus on introductory concepts and definitions.
- Safety considerations—The textbook continues to focus on real world safety issues adapted from actual case studies. For example, presented are an analysis of truck pick up tailgate cable failures, an unguarded in-running nip point of a treadmill belt drive, the sudden collapse of a cantilevered section of a walkway under demolition, the unexpected release of a hack squat machine carriage, the exposed shaft of an industrial finger jointer, and the safety of a two-wheeled, self-balancing electric personal transport.
- *Engineering material selection*—New charts from Professor Ashby's material selection procedure form the basis for introducing the student to selection and use of common engineering materials.
- *Elastic stability*—A design study presents an introduction to a captive cap that has two elastically stable positions: "opened" and "closed." The study references a patent and an exercise, the solution of which reveals a method of patent analysis.
- Abrasive wear—A device for screening seal materials is studied. An associated homework problem introduces the student to a patented test apparatus used to select the best materials for tricone drilling bits.
- *Impact absorber*—The concept of reducing impact forces and dissipating and dispersing impact loads is expanded to footwear with the introduction of a design study and homework problem involving a high heel shoe impact absorber.
- Fasteners—A design case study introduces a fastening device that is comprised of multiple components. A flexible slotted circular binding band in effect acts like a worm gear, and a thread screw member acts like a worm in driving the joined ends of the circular band. A new related homework problem provides the student with an opportunity to fully understand a related patent.
- Suspension system—A design study gives the student an opportunity to better understand a patented front and rear vehicle suspension system, and suspension related vehicle components.
- *Microscale and nanoscale machine elements*—A chapter contributed by Professor Cullinan presents a detailed study of micro–nano machine elements, such as mechanical sensors, actuators, and bearings that are used in microelectromechanical (MEMS) and nanoelectromechanical systems (NEMS).
- *Remote control racecars*—A design project case study authored by Professor Cullinan introduces readers to an exciting design-and-build racecar experience that involves design, analysis, experimentation, and evaluation, culminating with a student RC car race.
- *Appendix*—The appendices have been edited and condensed to focus on supplemental material that is most necessary for reference work with the textbook.

Part I

Although much of Part I of the text is a review of earlier courses, we would like to call attention to several particular sections.

- Sections 1.2, 1.3, and 1.4 deal with three of the broadest aspects of engineering—safety, ecology, and social significance. These are concerns to which today's students are particularly responsive.
- Section 1.7 presents a methodology for solving machine component problems. Embodied in this methodology is a sample problem format that includes a restatement, solution, and comments for the problem under the headings: known, find, schematic, decisions, assumptions, analysis, and comments. *Decisions* are choices made by the designer. Since design is an iterative decision-making process of synthesis, whenever the heading "decisions" is utilized, a design problem is presented. If a solution is presented without decisions being made, the problem is one of analysis. The inclusion of the category "decisions" allows the student to see clearly the difference between design and analysis. Once appropriate decisions have been made, analysis can follow. *Assumptions*, which are used in solving a problem, are statements about beliefs; for example, the material is homogeneous throughout. The design engineer and the student need to understand what assumptions are made in solving a problem. The listing of assumptions provides more opportunities for students of machine design to "think before doing." *Comments* present key aspects of the solution and discuss how better results might be obtained by making different design decisions, relaxing certain assumptions, and so on.
- Sections 1.8, 1.9, and 1.10 review fundamental energy relationships. Most students at this level need to gain insight and understanding concerning such basic matters as the relationship between work input to a rotating camshaft and work output at a translating follower, and the relationship between engine power, vehicle speed, and fuel consumption.
- Most teachers of Mechanical Engineering Design lament the weakness of their students in the area of free-body diagram analysis of loads. Unless the loading on a machine component is properly established, subsequent design or analysis is of little value. Section 2.2 is directed toward helping relieve this common deficiency and its associated problems.
- References are often an invaluable resource for the student as they provide in depth coverage of topics to which the text may only be able to devote a single paragraph. As such, *MIL-HDBK-17* is introduced to the student in Chapter 3 and *MIL-HDBK-5J* in Appendix F. These two references provide a wealth of pragmatic engineering knowledge regarding engineering materials and composites. The use of these volumes, along with the chapter references, has the ability to dramatically enhance a student's knowledge base.
- An elementary treatment of residual stresses is included in Chapter 4. An understanding of the basic concepts involved is vital to modern stress analysis, particularly when fatigue is present.
- Castigliano's method for determining elastic deflections and redundant reactions is included in Chapter 5. This method permits a ready solution to many problems not amenable to traditional elementary methods.
- Chapter 6 on Failure Theories, Safety Factors, Stress Intensity Factors, and Reliability includes introductory treatments of fracture mechanics and interference theory of statistical reliability prediction.
- Chapter 7 focuses on impact, which is also called shock, sudden, or impulsive loading.
- Chapter 8 contains a simplified, condensed, and introductory version of Fatigue Design and Fatigue Crack Growth. This chapter is particularly important, and represents primarily new material for most students.
- Chapter 9 deals with the various kinds of surface deterioration experienced by machine components. This is of great importance because more machine parts "fail" (cease to be suitable for performing their intended function) because of surface damage than from actual breakage.

Part II

Part II is concerned with the application of the fundamentals to specific machine components. In engineering practice, problems involving the design, analysis, or application of machine members can seldom be solved by applying the

fundamentals alone. As critically important as a knowledge of the underlying sciences is, it is seldom sufficient. Almost always some empirical information must be used, and good engineering judgment brought to bear. Actual engineering design problems seldom have only one correct answer. For example, engineering staffs of competing companies arrive at different product designs as solutions to the same problem. And these solutions change as new technology, new materials, new manufacturing methods, and new marketing conditions prevail. For many students, the course based on this text will provide their first experience in dealing with these kinds of professional engineering problems.

Most engineers find that this aspect of engineering adds to the interest and excitement of their profession. There is a close parallel between engineers and medical doctors in this respect: Both must solve real-life problems *now*, making full use of the best available scientific information. Engineers must design engines and build electronic apparatus even though scientists are still seeking a more complete knowledge of combustion and electricity. Similarly, medical doctors cannot tell their patients to await treatment until more research has been completed.

Even though the fundamentals treated in Part I are seldom *sufficient* for solving engineering problems relating to machine components, it is important that they be applied fully and consistently. In particular, a special effort has been made in Part II to deal with fatigue and surface considerations in a manner consistent with the treatment given in Chapters 8 and 9. This sometimes results in the development of procedures that vary in detail from those given in the specialized literature, but this discrepancy is not of major importance. What *is* of major importance is helping the student learn to approach engineering problems by applying the fundamentals and other scientific knowledge as extensively as possible, and then supplementing these with empirical data and judgment as required to get good solutions within available time limitations.

Few engineering schools allot sufficient time to cover all the machine components treated in Part II. In addition, many components are not treated in the book, and even more are not yet in existence. For these reasons, each component is treated not only as an end in itself, but also as a representative example of applying basic fundamentals and necessary empirical information to solve practical engineering problems.

Throughout Part II, the reader will find numerous instances in which ingenuity, insight, and imagination are called for in order to deal effectively with engineering problems associated with an individual machine component. The next step in the study of Mechanical Engineering Design usually involves the conception and design of a complete machine. As an introduction to this next step, the final two chapters of the book (Chapters 21 and 22), present (1) a case study of the design of the first commercially successful automotive automatic transmission, and (2) a case study of the design of remote control vehicles. These chapters can be found on the website for this text, http://www.wiley.com/college/juvinall. Here, as with numerous other designs of complete machines, one cannot help being impressed and inspired by the insights, ingenuity, and imagination (as well as the prolonged diligent effort) displayed by engineers and engineering students. Also illustrated in these case studies is the way that the design of any one component is often influenced by the design of related parts.

Because engineers will inevitably need to continue to deal with SI, British gravitational, and English engineering units, all three systems are used in the text and problems. Recalling the NASA/JPL Mars Climate Orbiter, where the root cause of the loss of the Orbiter spacecraft was the failed translation of English units into metric units in a segment of ground-base, navigation-related mission software, should help to remind the student just how important it is to understand and apply units properly.

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SYMBOLS

A	area, cross-sectional area, arm of planetary	C_R	reliability factor
	gear	c_{ρ}	volumetric specific heat
A	point A	$C_{\rm req}$	required value of C
A_0	original unloaded cross-sectional area	C_s	surface factor
a	influence coefficient	D	diameter, mean coil diameter, velocity factor
a, <i>a</i>	acceleration	d	diameter, major diameter, nominal diameter,
а	crack depth, radius of contact area of two		wire diameter
	spheres	d_{av}	average diameter
A_c	effective clamped area	d_b	diameter of base circle
$a_{\rm cr}$	critical crack depth	d_c	collar (or bearing) diameter
A_f	final area	dc/dN	crack propagation rate
A_r	area reduction	$(dc/dN)_o$	crack propagation rate at $(\Delta K)_o$
A_t	tensile stress area, tensile stress area of the	d_g	pitch diameter of gear
	thread	$d_{g} \ d_{i}$	minor diameter of the internal thread
\overline{B}	actual backlash	d_m	mean diameter
b	section width, half width of contact area	d_p	pitch diameter, pitch diameter of pinion
	measured perpendicular to axes of two	\dot{d}_r	root (or minor) diameter
	parallel contacting cylinders, gear face	E	modulus of elasticity, elastic proportionality
	width, band width		constant, tensile elastic modulus
C	spring index, overall heat transfer	E	modulus of elasticity (tension)
	coefficient, rated load capacity, heat transfer	E_p	plastic strain
	coefficient, constant (material property)	e	distance between the neutral axis and the
C	specific heat		centroidal axis, efficiency, eccentricity, train
C	distance from the neutral axis to the extreme		value, edge distance for joint, percent
	fiber, half of crack length, radial clearance,		elongation at break
	center distance, distance between shafts,	e/D	edge margin
	crack length	E_b	Young's modulus for the bolt
\overline{c}	distance from the centroidal axis to the	E_c	Young's modulus for clamped member,
	extreme inner fiber, actual distance between		compression modulus of elasticity
	gear and pinion centers	E_s	secant modulus
$c_{\rm cr}$	critical crack length	E_t	tangent modulus
CR	contact ratio	F	force, compressive force between the
\overline{CR}	actual contact ratio		surfaces
CG	center of gravity	f	relative hardenability effectiveness,
C_G	gradient factor or gradient constant		coefficient of friction
c_{i}	distance from the neutral axis to the extreme	\mathbf{F}, F	force
	inner fiber	F_a	axial force
C_L	load factor	F_b	bolt axial load
$C_{ m Li}$	life factor	$F_{ m bru}$	bearing ultimate strength
c_o	distance from the neutral axis to the extreme	$F_{\rm bry}$	bearing yield strength
	outer fiber	F_c	clamping force
CP	center of aerodynamic pressure	f_c	collar (or bearing) coefficient of friction
C_p	elastic coefficient	F_d	drag force, dynamic load

Г			
$F_{\rm cy}$	compression yield strength	$K_{\mathrm{I}c}$	critical stress intensity factor for tensile
F_{e}	equivalent radial load, equivalent static		loading (mode I)
	force, external force	K_a	application factor
$\mathbf{F}_{\mathrm{ext}}$	external force vector applied on a member	K_B	constant of proportionality
$F_{\rm ga}$	gear axial force	k_b	spring constant for the bolt
$F_{\rm gr}$	gear radial force	K_c	fracture toughness or critical stress intensity
$F_{ m gr}^{ m gr}$ $F_{ m gt}$ F_{i}	gear tangential force		factor
${F}_{i}$	initial tensile force, initial clamping force	k_c	spring constant for clamped members
\mathbf{F}_{int}	internal force vector at a cross-section	KE	kinetic energy
F_n	normal force	K_f	fatigue stress concentration factor
f_n	natural frequency	K_i	curvature factor for inner fiber, effective
F_r	radial load, radial force		stress concentration factor for impact
F_s	strength capacity		loading, constant used for calculating initial
$F_{ m solid}$	force when solid		bolt-tightening force
$F_{ m su}$	shear ultimate strength	K_m	mounting factor
\vec{F}_t	thrust force, tendon force, tangential force,	K_{\max}	stress intensity factor at $\sigma_{\rm max}$
	thrust load	K_{\min}	stress intensity factor at σ_{\min}
$F_{\rm tu}$	tensile ultimate strength	$k_{ m ms}$	mean stress factor
	tensile yield strength	K_o	curvature factor for outer fiber, overload
$F_{ m ty} = F_{ m w}$	wear capacity	U	factor, critical stress intensity factor for
$F_{\rm wa}^{''}$	worm axial force		infinite plate with central crack in uniaxial
$F_{\rm wr}^{\rm wa}$	worm radial force		tension
$F_{\mathrm{wt}}^{\mathrm{wi}}$	worm tangential force	K_r	life adjustment reliability factor
$\overset{\text{wt}}{G}$	torsional or shear modulus of elasticity	$k_r^{'}$	reliability factor
g	gravitational acceleration or acceleration of	$K_{s}^{'}$	stress concentration factor for static loading
O	gravity, grip length	K_{t}^{3}	theoretical or geometric stress concentration
g_c	constant of proportionality,	ı	factor
00	32.2 lbm-ft/lb-s ²	k_t	temperature factor
H	surface hardness, time rate of heat	K_v	velocity or dynamic factor
	dissipation	K_w	Wahl factor, material and geometry factor
h	section depth, height of fall, leg length, weld	$\stackrel{\scriptstyle L}{L}$	length, contact length measured parallel to
	size, film thickness, height		the axis of contacting cylinder, lead, length
h_0	minimum film thickness		of weld, life corresponding to radial load F_r ,
H_B	Brinell hardness number		or life required by the application, pitch cone
I	polar moment of inertia, moment of inertia,		length
•	geometry factor, stress invariant	L_0	original unloaded length
i	integer	L_e^{-0}	equivalent length
$I_{_{X}}$	moment of inertia about x axis	L_f	final length, free length
J^{x}	polar moment of inertia, spur gear geometry	L_R	life corresponding to rated capacity
Ü	factor	L_s	solid height
K	curvature factor, spring rate for angular	L, ST, LT	longitudinal direction, short transverse
**	deflection, stress intensity factor, wear	2,51,21	direction, long transverse direction
	coefficient	M	moment, internal bending moment, bending
k	spring rate, thermal conductivity, spring rate	171	moment
K	for linear deflection, number of standard	M_0	redundant moment
	deviations, shaft spring rate	m	mass, strain-hardening exponent, module
K	thermal conductivity	"	(used only with SI or metric units)
K'	section property	m'	mass per unit length of belt
$K_{\rm I}$	stress intensity factor for tensile loading	$\mathbf{M}_{\mathrm{ext}}$	
M	(mode I)	M_f	external moment vector applied on a member moment of friction forces
	(mode 1)	N_f	moment of friction forces

$\mathbf{M}_{\mathrm{int}}$	generalized internal moment vector at a	<u>r</u>	radius, reliability
1.7	cross-section	\overline{r}	radial distance to the centroidal axis
M_n	moment of normal forces	$r_{a(\max)}$	maximum noninterfering addendum circle
N	fatigue life, total normal load, number of		radius of pinion or gear
	active coil turns, number of teeth, number of	$r_{\rm ag}^{\rm max}$	maximum allowable addendum radius on the
	friction interfaces, number of cycles		gear to avoid interference
n	rotating speed, number of cycles, normal	$r_{\rm ap}^{\rm max}$	maximum allowable addendum radius on the
	force, number of equally spaced planet		pinion to avoid interference
	gears, index (subscript), Ramberg-Osgood	$r_{\rm ap}, r_{\rm ag}$	addendum radii of the mating pinion and
N'	parameter virtual number of teeth		gear
N.A.	neutral axis	r_b	base circle radius, back cone radius
	critical speed	$r_{\rm bp}, r_{\rm bg}$	base circle radii of the mating pinion and
n_c	number of teeth	1 0	gear
$\frac{N_e}{N_t}$	total number of turns, number of teeth in the	r_c	chordal radius
$i\mathbf{v}_t$	sprocket		friction radius
P	load, cumulative probability of failure,	$rac{r_f}{\overline{r}_g}$	actual pitch radius of gear
1	bearing unit load, average film pressure,	r:	inner radius
	radial load per unit of projected bearing area,	$\frac{r_i}{\overline{r}_p}$	actual pitch radius of pinion
	pitch point, diametral pitch (used only with	R_m^p	modulus of resilience
	English units), diameter or number of teeth	r_n	radial distance to the neutral axis
	of planet, band force, load (force), uniform	r_o^n	outer radius
	load	$\overset{o}{S}$	linear displacement, total rubbing distance,
\overline{P}	actual pitch		Saybolt viscometer measurement in seconds,
p	frequency of occurrence, probability of		bearing characteristic number or
Р	failure, surface interface pressure, pitch, film		Sommerfeld variable, diameter or number of
	pressure, circular pitch, uniform level of		teeth of sun gear, slip
	interface pressure, pressure	$S_{\rm cr}$	critical unit load
\overline{p}	actual circular pitch	S_e	elastic limit
p_0	maximum contact pressure	$S_{ m eq}$	equivalent stress—see Table F.4
p_a	axial pitch	$S\dot{F}$	safety factor
p_b	base pitch	$S_{ m fe}$	surface fatigue strength
P_c	tension created by centrifugal force	S_H	surface endurance strength
$P_{\rm cr}$	critical load	S-N	fatigue stress versus cycles
PE	potential energy	$S_{ m max}$	maximum fatigue cycle stress—see Table F.4
p_{max}	allowable pressure, maximum normal	S_n	endurance limit
	pressure	$S'_n \\ S_p$	standard fatigue strength for rotating bending
p_n	circular pitch measured in a plane normal to	S_p	proof load (strength)
	the teeth	$S_{\mathrm{sy}} \\ S_{u}$	shear yield strength
Q	heat energy transferred to the system, load,		ultimate strength, ultimate tensile strength
	total tangential force, flow rate, mass flow	$S_{\rm uc}$	ultimate strength in compression
	rate	$S_{ m us}$	ultimate shear strength, ultimate torsional
q	number of revolutions, notch sensitivity	G.	shear strength
	factor, tangential force	$S_{ m ut}$	ultimate strength in tension
Q_f	volume of lubricant per-unit time flowing	S_{y}	yield strength
	across	$rac{S_{ m yc}}{S_{ m yt}}$	yield strength in compression
Q_s	side leakage rate	$S_{ m yt}$	yield strength in tension
R	radius, transmission speed ratio, area ratio,	T	torque, brake torque, band brake torque
	radius of curvature, diameter or number of	t T	time, thickness, nut thickness, throat length
	teeth of ring or annulus gear, ratio of gear	T_a	alternating torque
	and pinion diameter, load ratio, fatigue cycle	t_a	air temperature, ambient air temperature
	stress ratio	T_e	equivalent static torque

T_f	friction torque	α_2	normalized crack size at c_2
T_m^j	modulus of toughness, mean torque	α_{n}	thread angle measured in the normal
t_o	average oil film temperature, oil temperature	α_n	plane
t_s	average temperature of heat-dissipating	Δ	deflection, material parameter important in
LS	surfaces	A	computing contact stress
U	stored elastic energy, impact kinetic energy,	δ, δ	deflection
U	laminar flow velocity	δ , δ	linear deflection, wear depth
U'		ΔA	
	complementary energy		change in total arrays of the southern
V	internal transverse shear force, shear force,	ΔE	change in total energy of the system
1 7 17	volume	ΔKE	change in kinetic energy of the system
\mathbf{V}, V	linear velocity, gear pitch line velocity	ΔK	stress intensity range
U	velocity at impact, sliding velocity	ΔK_o	stress intensity range at the point o
V_{60}	cutting speed in feet per minute for 60-min	ΔL	change in length
	tool life under standard cutting conditions	ΔPE	change in gravitational potential energy of
$V_{ m av}$	average velocity		the system
V_g	gear tangential velocity, pitch line velocity	ΔN_{12}	number of cycles during crack growth from
	of the gear		c_1 to c_2
$V_{ m gt}$	velocity of gear at contact point in tangent	$\delta_{\scriptscriptstyle S}$	solid deflection
	direction	$\delta_{ m st}$	deflection caused by static loading (static
$V_{ m pt}$	velocity of pinion at contact point in tangent		deflection)
	direction	ΔT	temperature change
$V_{ m gn}$	velocity of gear at contact point in normal	ΔU	change in internal energy of the system
Σ,	direction	λ	lead angle, helix angle, ratio of actual to
$V_{ m pn}$	velocity of pinion at contact point in normal		ideal distance between gear and pinion
	direction		centers
$V_{_S}$	sliding velocity	ϕ	angle between the principal axes and the x
V_w	worm tangential velocity		and y axes, angle giving position of
\overline{W}	work done, weight, volume of material worn		minimum film thickness, pressure angle,
	away, total axial load		angle of wrap
\dot{W}	power	ϕ_n	pressure angle measured in a plane normal to
W	load, load intensity, gravitational force,		the teeth
	width	$\overline{\phi}$	actual pressure angle
Y	Lewis form factor based on diametral pitch	γ	pitch cone angle
	or module, configuration factor	$\gamma_{xy}, \gamma_{xz}, \gamma_{yz}$	shear strains
v	distance from the neutral axis, Lewis form		mean, viscosity
J	factor	μ μ	Poisson's ratio—see Appendix F
$Y_{\rm cr}$	configuration factor at critical crack size	V	Poisson's ratio
Z	section modulus	ϵ	normal strain
_			principal strains
		$\epsilon_1, \epsilon_2, \epsilon_3$	
Greek Let	ters	ϵ_f	strain at fracture
		ϵ_p	plastic strain
α	angular acceleration, coefficient of thermal	ϵ_T	"true" normal strain
	expansion, angles measured clockwise	$\epsilon_{T\!f}$	true normal strain at fracture
	positive from the 0° gage to the principal	$\epsilon_x, \epsilon_y, \epsilon_z$	normal strains
	strain axes numbers 1 and 2, factor by which	θ	angular displacement, angular deflection,
	the compressive strength is reduced through		slope

 $\theta_{P_{\rm max}}$

position of maximum film pressure

normal stress, standard deviation, uniform

mass density, radial distance

uniaxial tensile stress

buckling tendencies, thread angle, contact

angle, cone angle, normalized crack size

normalized critical crack size

normalized crack size at c_1

 $\alpha_{\rm cr}$

 α_1