

# Mechanics of Engineering Materials

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second edition

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## Preface to the Second Edition

The historical development of this text was outlined in the Preface to the previous edition. During the nine years since its publication there have been no major changes in the theory of the subject but there have been advances in some of the manipulative tools which can be used to solve problems related to the strength of materials. In addition, with experience and hindsight one can always improve the presentation of material to make it more digestible to the reader.

In this new edition we have engaged the services of a new author Dr Cecil Armstrong to assist in a complete update of the text. New features of the book include the use of versatile solution techniques based on spreadsheets, the use of colour to enhance diagrams and provide emphasis of key concepts, an introduction to matrix methods as a powerful tool in engineering analysis, and the incorporation of additional practical problems to illustrate the application of the theory.

The essential structure and content of Chapters 1–9 has not been changed although regular users of the previous text will see subtle changes in the mode of presentation to make this important fundamental information more meaningful to students. Chapter 10 from the original text has been removed because it was felt that modern computer programs provide powerful tools for the solutions of problems in structures. In a general text of this type it is impossible to do justice to the wide range of methods available to analyse structures. Hence, we have moved the important fundamental information on tension coefficients, energy methods, etc. to other

chapters and omitted the superficial and to some extent outdated solution methods based on scale drawings.

The remaining chapters have benefited from a complete overhaul and the chapter on Finite Elements has been almost totally re-written. Throughout the text the reader is introduced to the concept of creating a spreadsheet representation of a problem in strength of materials. Once such a solution has been formulated it is then a relatively simple matter to study the effects of changes in variables, to plot graphs of key variables and to optimize solutions. This is shown to be a very powerful solution methodology which provides a completely new insight to each problem and open up a whole new approach to problem solving in many subject areas.

The authors derived a great deal of pleasure in working on this new version of the text. However, although he participated fully in the revision of the book, regrettably Peter Benham died before he was able to see the new book in print. We know that the many young people who studied under him, and the readers all over the world who benefited from his books, were greatly saddened by his death. It is our hope that this new edition, about which he was very excited and proud, will be a fitting tribute to his memory.

R. J. CRAWFORD C. G. ARMSTRONG 10 November 1995

# Preface to the First Edition

The S.I. edition of Mechanics of Solids and Structures by P. P. Benham and F. V. Warnock was first published in 1973. It appears to have been very well received over the ensuing period. This preface is therefore written both for those who are familiar with the past text and also those who are approaching this subject for the first time. Although the subject matter is still basically the same today as it has been for decades, there are a few developing topics which have been introduced into undergraduate courses such as finite element analysis, fracture mechanics and fibre composite materials. In addition, style of presentation and illustrations in engineering texts have changed for the better and certain limitations of the previous edition, e.g. the number of problems and worked examples, needed to be rectified. Professor Warnock died in 1976 after a period of happy retirement and so it was left to the other author and the publisher to take the initiative to construct a new textbook.

In order to provide fresh thinking and reduce the time of rewriting Dr Roy Crawford, Reader in Mechanical Engineering at the Queen's University of Belfast, was invited, and kindly agreed, to join the project as a co-author. Although the present text might be regarded as a further edition of the original book the new authorship team preferred to make a completely fresh start. This is reflected in the change of title which is widely used as an alternative to Mechanics of Solids. The dropping of the reference to structures does not imply any reduction in that topic as will be seen in the contents.

In order that the book should not become any larger with the proposed expansion of material in some chapters, it was decided that the previous three chapters on experimental stress analysis should be omitted as there are several excellent texts in this field. The retention of that part of the book dealing with mechanical properties of materials for design was regarded as important even though there are also specialized texts in this area.

The main part (Ch. 1–18) of this new book of course still deals with the basic subject of Solid Mechanics, or Mechanics of Materials, whichever title one may prefer, being the study of equilibrium and displacement systems in engineering components and structures to enable designs to be effected in terms of stress and strain and the selection of materials. These eighteen chapters cover virtually all that is required in the three-year

syllabus of a university or polytechnic degree course in engineering, or the examinations of the engineering Council, C.N.A.A. etc.

Although there is a fairly natural ordering of the material there is some scope for variation and lecturers will have their own particular detailed preferences.

As in the previous text, the first eleven chapters are concerned with forces and displacements in statically-determinate and indeterminate components and structures, and the analysis of uniaxial stress and strain due to various forms of loading such as bending, torsion, pressure and temperature change. The basic concepts of strain energy (Ch. 9) and elastic stability (Ch. 11) are also introduced. In Chapter 12 a study is made of two-dimensional states of stress and strain with special emphasis on principal stresses and the analysis of strain measurements using strain gauges. Chapter 13 combines two chapters of the previous book and brings together the topics of yield prediction and stress concentration which are of such importance in design.

Also included in these two chapters is an elementary introduction to the stress analysis and failure of fibre composite materials. These relatively new advanced structural materials are becoming increasingly used, particularly in the aerospace industry, and it is essential for engineers to receive a basic introduction to them. These thirteen chapters constitute the bulk of the syllabuses covered in first and second-year courses.

Four of the next five chapters appeared in the previous text and deal with more advanced or specialized topics such as thick-walled pressure vessels, rotors, thin plates and shells and post-yield or plastic behaviour, which will probably occupy part of final-year courses.

One essential new addition is an introductory chapter on finite element analysis. It may seem presumptuous even to attempt an introduction to such a broad subject in one chapter, but it is an attempt to provide initial encouragement and confidence to proceed to the complete texts on finite elements.

Chapters 19 to 22 cover much the same ground as in the previous text, but have been brought up to date particularly in relation to fracture mechanics. Since these chapters have such importance in relation to design, a number of worked examples have been introduced, together with problems at the end of each chapter. Bibliographies have still been included for further reading as required. The first Appendix covers the essential material on properties of areas. The second deals with the simple principles of matrix algebra. A useful table of mechanical properties is provided in the third Appendix.

One of the recommendations of the Finniston Report to higher education was that theory should be backed up by more practical industrial applications. In this context the authors have attempted to incorporate into the worked examples and problems at the end of each chapter realistic engineering situations apart from the conventional examination-type applications of theory.

There had been a number of enquiries for a solutions manual for the previous text and this can be very helpful to both lecturer and student. Consequently this text is accompanied by another volume which contains worked solutions to nearly 300 problems. The manual should be used alongside the main text, so that steps in each solution can be referred back to the appropriate

development in the relevant chapter. It is most important not to approach solutions on the basis of plucking the 'appropriate formula' out of the text, inserting the numbers, and manipulating a calculator!

Every effort has been made by the authors to ensure accuracy of text and solutions, but lengthy experience demonstrates human fallibility in this respect. When errors subsequently come to light they will be corrected at the next reprinting and readers' patience and comments will be appreciated!

Some use has been made of data and diagrams from other published literature and, in addition to the individual references, the authors wish to make grateful acknowledgement to all persons and organizations concerned.

P. P. BENHAM R. J. CRAWFORD 1987

## Notation

distance

```
angle, coefficient of thermal expansion
\alpha
\beta
       shear strain, surface energy per unit area
\gamma
δ
       deflection, displacement
       direct strain
\epsilon
       efficiency, viscosity
\eta
       angle, angle of twist, co-ordinate
À
       lack of fit
       Poisson's ratio
\nu
       radius of curvature, density
ρ
\sigma
       direct stress
       shear stress
\tau
       angle, co-ordinate, stress function
       angular velocity
A
\boldsymbol{C}
       complementary energy
\boldsymbol{D}
       diameter
       Young's modulus of elasticity
\boldsymbol{E}
\boldsymbol{\mathit{F}}
       shear or rigidity modulus of elasticity, strain energy release rate
\boldsymbol{G}
H
       second moment of area, product moment of area
Ι
\mathcal{J}
       polar second moment of area
       bulk modulus of elasticity, fatigue strength factor, stress concentration
K
       factor, stress intensity factor
L
       length
M
       bending moment
       number of stress cycles, speed of rotation
N
P
       force
       shear force
Q
R
       force, radius of curvature, stress ratio
S
       cyclic stress
 T
       temperature, torque
 \boldsymbol{U}
       strain energy
 V
       volume
 W
       weight, load
 X
       body force
 Y
       body force
\boldsymbol{Z}
       body force, section modulus
       area, distance, crack length
a
       breadth, distance, crack length
 b
С
       distance
 d
       depth, diameter
       eccentricity, base of Napierian logarithms
 е
       gravitational constant
```

NOTATION xiii

- j number of joints
- k diameter ratio of cylinder
- l length
- m mass, modular ratio, number of members
- n number
- p pressure
- shear flow
- r co-ordinate, radius, radius of gyration
- s length
- t thickness, time
- u displacement in the x- or r-direction
- v deflection, displacement in the y- or  $\theta$ -direction, velocity
- w displacement in the z-direction, load intensity
- x co-ordinate, distance
- y co-ordinate, distance
- z co-ordinate, distance

It should be noted that a number of these symbols have also been used to denote constants in various equations.

# Contents

Preface to the second edition  Preface to the first edition  Notation		ix		9 Torsion of a thin circular tube 9 Joints 9 Summary 9 Problems	
		x xii	2.10		
				Tionin	60
			3 St	ress-Strain Relations	64
			3.1	Deformation	64
		_	3.2	Strain	64
1 S	tatically Determinate Force Systems	1		Elastic load-deformation behaviour of material	
1.1	Revision of statics	1		Elastic stress-strain behaviour of materials	67
1.2	Resultant force and moment	5	3.5	Lateral strain and Poisson's ratio	70
1.3	Types of structural and solid-body components	6	3.6	Thermal strain	70
	Types of support and connection for structural			General stress-strain relationships	71
	components	7		Strains in a statically determinate problem	74
1.5	Statical determinacy	9		Elastic strain energy	75
1.6	Free-body diagrams	9		Strain energy from normal stress	75
	Determinacy criteria for structures	10		Strain energy from shear stress	76
1.8	Determination of axial forces by equilibrium			Plastic stress-strain behaviour of materials	77
	statements	12	3.13	Viscoelastic stress-strain behaviour of materials	
1.9	Forces by the method of tension coefficients	18		Summary	79
1.10	Bending of slender members	21		Problems	79
	Force and moment equilibrium during bending	23			
1.12	Sign convention for bending	23	4 St	atically Indeterminate Stress Systems	82
1.13	Shear-force and bending-moment diagrams	24		·	
1.14	Cantilever carrying a concentrated load	24	4.1	Interaction between components of different	82
1.15	Cantilever carrying a uniformly distributed load	25	4.2	stiffness	88
1.16	Simply supported beam carrying a uniformly			Restraint of thermal strain	91
	distributed load	26		Volume changes	93
1.17	Point of contraflexure	28		Constrained material	93
1.18	Torsion of members	28	4.3	Maximum stress due to a suddenly applied	94
1.19	Members subjected to axial force, bending		1.6	load Maximum atmass days to immed	95
	moment and torque	29	7.0	Maximum stress due to impact	97
	Torque diagram	30		Summary Problems	97
1.21	The principle of superposition	31		Froncins	71
	Summary	32			
	Problems	33	5 T	orsion	102
			5.1	Torsion of a thin-walled cylinder	102
			5.2	Torsion of a solid circular shaft	103
2 S	tatically Determinate Stress Systems	43	5.3	Relation between stress, strain and angle of	
2.1	Stress: normal, shear and hydrostatic	43		twist	104
	A statically determinate stress system	45	5.4	Relation between torque and shear stress	104
	Assumptions and approximations	46	5.5	The torsion relationship	105
	Tie bar and strut (or column)	46		Torsion of a hollow circular shaft	106
	Suspension-bridge cables	50		Torsion of non-uniform and composite shafts	108
	Thin ring or cylinder rotating	51	5.8	Torsion of a thin tube of non-circular section	112
	Thin shells under pressure	52		Torsion of thin rectangular strip	115
	Shear in a coupling	57	5.10	Effect of warping during torsion	118

vi CONTENTS

5.11	Torsion of solid rectangular and square		9.3	Elastic strain energy: normal stress and	224
	cross-sections	119		shear stress	236
	Summary	119		Strain energy in torsion	236
	References	120		Strain energy in bending	237
	Problems	120		Helical springs	240
				Shear deflection of beams	243
6 Be	ending: Stress	125		Virtual work	245
	Shear-force and bending-moment			Displacements by the virtual work method	246
0.1	distributions	125		Strain energy solutions for forces	248
6.2	Relationships between loading, shear force		9.11	Complementary energy solution for	240
0.2	and bending moment	130	0.12	deflections	249 252
6.3	Stress distribution in pure bending	132		Bending deflection of beams	257
	Deformation in pure bending	133	9.13	The reciprocal theorem	259
	Stress-strain relationship	134		Summary	259
	Equilibrium of forces and moments	135		Problems	239
	The bending relationship	136			
	The general case of bending	136	10 H	Buckling Instability	264
	Section modulus	137	10.1	Stability of equilibrium	264
6.10	Beams made of dissimilar materials	139	10.2	Buckling characteristics for real struts	268
	Combined bending and end loading	147	10.3	Eccentric loading of slender columns	274
	Eccentric end loading	148	10.4	Struts having initial curvature	276
6.13	Asymmetrical or skew bending	151	10.5	Empirical formulae for design	279
	Shear stresses in bending	156	10.6	Buckling under combined compression	
6.15	Bending and shear stresses in I-section			and transverse loading	280
	beams	160	10.7	Pin-ended strut carrying a uniformly	
6.16	Shear stress in thin-walled open sections			distributed lateral load	281
	and shear centre	164	10.8	Other examples of instability	284
6.17	Bending of initially curved bars	168	10.9	Local buckling	285
6.18	Beams with a small radius of curvature	196		Summary	287
	Summary	173		References	287
	Reference	173		Problems	287
	Problems	173			
7 D	I' CI I D Gerden	105	11 5	Stress and Strain Transformations	292
/ Be	ending: Slope and Deflection	185	11.1	Symbols, signs and elements	292
7.1	The curvature-bending-moment relationship	185	11.2	Stresses on a plane inclined to the direction	
7.2	Slope and deflection by the double-integration			of loading	293
	method	186	11.3	Element subjected to normal stresses	294
	Discontinuous loading: Macaulay's method	193	11.4	Element subjected to shear stresses	295
	Superposition method	202	11.5	Element subjected to general two-dimensional	
	Deflections due to asymmetrical bending	204		stress system	296
7.6	Beams of uniform strength	205	11.6	Mohr's stress circle	298
	Summary	206	11.7	Principal stresses and planes	301
	Problems	207	11.8	Maximum shear stress in terms of principal	
				stresses	302
8 St	atically Indeterminate Beams	211	11.9	General two-dimensional state of stress at	
8.1	Double-integration method	211		a point	302
	Superposition method	216		Maximum shear stress in three dimensions	306
	Moment-area method	220		States of strain	307
	Summary	232		Normal strain in terms of co-ordinate strains	307
	Problems	232		Shear strain in terms of co-ordinate strains	308
				Mohr's strain circle	310
Q F	nergy Methods	235		Principal strain and maximum shear strain	310
				Experimental stress analysis	311
	Work done by a single load	235	11.17	Rosette strain computation and circle	21.4
9.2	Work done by a single moment	235		construction	314

11 10	6 11 . 1 6	217	14.11	Changes in a nation of varying thickness with	
	Spreadsheet solution for strain gauge rosette	317 319	14.11	Stresses in a rotor of varying thickness with rim loading	403
	Relationships between the elastic constants Stress and strain transformations in composites			Summary	407
		322		Problems	407
	Analysis of a lamina Analysis of a laminate	329		1 TOOLETTS	107
	In-plane behaviour of a symmetric laminate	330			
	Flexural behaviour of a symmetric laminate	334	15 E	I Disatisias	413
11.27	Summary	334	15 E	lementary Plasticity	
	References	335	15.1	Plastic bending of beams: plastic moment	413
	Problems	335		Plastic collapse of beams	418
	Trongins	000		Plastic torsion of shafts: plastic torque	423
12 \	ield Criteria and Stress Concentration	340	15.4	Plasticity in a pressurized thick-walled cylinder	
12.1	Yield criteria: ductile materials	340		Plasticity in a rotating thin disc	429
12.1	Fracture criteria: brittle materials	348	15.6	Residual stress distributions after plastic	423
12.2		350		deformation	431
12.3	Strength of laminates Concepts of stress concentration	351		Summary	436
12.5	Concentrated loads and contact stresses	353		Problems	436
12.5	Geometrical discontinuities	356			
12.7	Yield and plastic stress concentration factors	362			
12.7	Summary	363	16 T	Thin Plates and Shells	439
	References	363	16.1	Assumptions for small deflection of thin	
	Problems	363		plates	439
	Tionems	202	16.2	Relationships between moments and	
13 N	Variation of Stress and Strain	367		curvatures for pure bending	439
			16.3	Relationships between bending moment	
13.1	Equilibrium equations: plane stress Cartesian co-ordinates	367		and bending stress	444
12.2	Equilibrium equations: plane stress	307	16.4	Relationships between load, shear force and	
13.2	cylindrical co-ordinates	368		bending moment	444
13.3	Strain in terms of displacement: Cartesian	300	16.5	Relationships between deflection, slope and	
13.3	co-ordinates	370		loading	445
13.4	Strain in terms of displacement: cylindrical	370	16.6	Plate subjected to uniform pressure	446
13.7	co-ordinates	371	16.7	Plate with central circular hole	448
13.5	Compatibility equations: Cartesian	371	16.8	Solid plate with central concentrated force	450
15.5	co-ordinates	373	16.9	Other forms of loading and boundary	
13.6	Compatibility equations: cylindrical			condition	451
	co-ordinates	374		Axi-symmetrical thin shells	453
13.7	Equilibrium in terms of displacement:			Local bending stresses in thin shells	454
	plane stress Cartesian co-ordinates	377	16.12	Bending in a cylindrical storage tank	458
	Summary	377		Summary	459
	Problems	378		Bibliography	460
				Problems	460
14 A	Applications of the Equilibrium and				
	Strain-Displacement Relationships	379			1/2
14.1	Boundary conditions	379	17 F	Finite Element Method	463
14.2	Shear stress in a beam	380	17.1	Principle of finite element method	463
14.3	Transverse normal stress in a beam	381	17.2	Analysis of uniaxial bars	464
14.4	Stress distribution in a pressurized	001	17.3	Analysis of frameworks	469
	thick-walled cylinder	382	17.4	Analysis of beam elements	476
14.5	Methods of containing high pressure	390	17.5	Analysis of continua	483
14.6	Stresses set up by a shrink-fit assembly	391	17.6	Stiffness matrix for a triangular element	484
14.7	Stress distribution in a pressurized		17.7	Effect of mesh density	488
	compound cylinder	392	17.8	Other continuum elements	489
14.8	Spreadsheet solution for stress in a			Summary	490
	compound cylinder	395		Reference	490
14.9	Stress distribution in a thin rotating disc	397		Bibliography	490
14.10	Rotational speed for initial vielding	401		Problems	491

	Tension, Compression, Torsion and	493	20.5 Fracture mechanics for fatigue 20.6 Influential factors	554 558
	Hardness	493	20.7 Cumulative damage	563
18.1	Stress-strain response in a uniaxial tension		20.8 Failure under multi-axial cyclic stresses	564
	test	493	20.9 Fatigue of plastics and composites	567
18.2	Stress-strain response in a compression test	497	Summary	568
18.3	Stress-strain response in a torsion test	498	References	569
18.4	Plastic overstrain and hysteresis	500	Bibliography	569
18.5	Hardness measurement	500	Problems	569
18.6	Hardness of viscoelastic materials	506		
18.7	Factors influencing strength, ductility and		21 Cross and Viscoslasticity	573
	hardness	506	21 Creep and Viscoelasticity	
	Summary	510	21.1 Stress-strain-time-temperature relationships	573
	Reference	511	21.2 Empirical representations of creep behaviour	574
	Bibliography	511	21.3 Creep-rupture testing	577
	Problems	512	21.4 Tension creep test equipment	578
			21.5 Creep during pure bending of a beam	578
19 Fracture Mechanics		514	21.6 Creep under multi-axial stresses	580
			21.7 Stress relaxation	582
19.1	Fracture concepts	514	21.8 Stress relaxation in a bolt	583
19.2	Linear-elastic fracture mechanics	515	21.9 Creep during variable load or temperature	584
19.3	Strain energy release rate	516	21.10 Creep-fatigue interaction	586
19.4	Stress intensity factor	518	21.11 Viscoelasticity	587
19.5	Modes of crack tip deformation	521	21.12 Creep behaviour of plastics	588
19.6	Experimental determination of critical	F21	21.13 Designing for creep in plastics	590
	stress intensity factor	521	21.14 Creep rupture of plastics	593
19.7	Fracture mechanics for ductile materials	525	Summary	593
19.8	Toughness measurement by impact testing	534	References	594
19.9	Relationship between impact testing and		Bibliography	594
	fracture mechanics	535	Problems	594
	Summary	537		
	References	537		
	Bibliography	538	Appendix A: Properties of Areas	598
	Problems	538		
			Appendix B: Introduction to Matrix Algebra	609
20 1	Fatigue	544		
20.1	Forms of stress cycle	544	Appendix C: Table of Mechanical Properties of	
20.1	Test methods	5 <del>44</del> 546	Engineering Materials	613
20.2	Fatigue data	5 <del>4</del> 8	Act II D. A Devillen	614
20.3	Micromechanisms of fatigue: initiation and	טדע	Appendix D: Answers to Problems	614
20.4		552	Index	624
	propagation	334	I HUCA	ULT

# Statically Determinate Force Systems

Structural and solid-body mechanics are concerned with analysing the effects of applied loads. These are *external* to the material of the structure or body and result in *internal* reacting forces, together with deformations and displacements, conforming to the principles of Newtonian mechanics. Hence a familiarity with the principles of statics, the cornerstone of which is the concept of *equilibrium of forces*, is essential.

A force system is said to be statically determinate if the internal forces can be calculated by considering only the forces acting on the system.

Forces result in four basic forms of deformation or displacement of structures or solid bodies and these are tension, compression, bending, and twisting.

The equilibrium conditions in these situations are discussed so that the forces may be determined for simple engineering examples.

#### 1.1 Revision of statics

A particle is in a state of equilibrium if the resultant force and moment acting on it are zero. This hypothesis can be extended to clusters of particles that interact with each other with equal and opposite forces but have no overall resultant. Thus it is evident that solid bodies, structures, or any subdivided part, will be in equilibrium if the resultant of all external forces and the resultant of all moments are zero. This may be expressed mathematically in the following six equations which relate to Cartesian coordinate axes x, y and z.

$$\left.\begin{array}{l}
\Sigma F_x = 0 \\
\Sigma F_y = 0 \\
\Sigma F_z = 0
\end{array}\right\}$$
[1.1]

where  $F_x$ ,  $F_y$  and  $F_z$  represent the components of force vectors in the coordinate directions.

$$\left.\begin{array}{l}
\Sigma M_x = 0 \\
\Sigma M_y = 0 \\
\Sigma M_z = 0
\end{array}\right\}$$
[1.2]

where  $M_x$ ,  $M_y$  and  $M_z$  are components of moment vectors caused by the external forces acting about the axes x, y, z.

The above six equations are the necessary and sufficient conditions for equilibrium of a body.

If the forces all act in one plane, say z = 0, then

$$\Sigma F_z = \Sigma M_x = \Sigma M_y = 0$$

are automatically satisfied and the equilibrium conditions to be satisfied in a two-dimensional system are

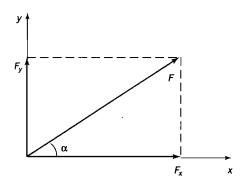
$$\left.\begin{array}{l}
\Sigma F_x = 0 \\
\Sigma F_y = 0 \\
\Sigma M_z = 0
\end{array}\right\}$$
[1.3]

Forces and moments are vector quantities and may be resolved into components; that is to say, a force or a moment of a certain magnitude and direction may be replaced and exactly represented by two or more components of different magnitudes and in different directions.

Considering firstly the two-dimensional case shown in Fig. 1.1, the force F may be replaced by the two components  $F_x$  and  $F_y$  provided that

$$\begin{cases}
F_x = F \cos \alpha \\
F_y = F \sin \alpha
\end{cases}$$
[1.4]

Fig. 1.1



Note that throughout this book, externally applied forces will be shown coloured. Internal forces, that is those within a structural element, will be shown in black.

If the force F were arbitrarily oriented with respect to three axes x, y, z as in Fig. 1.2, then it could be replaced or represented by the following components:

$$\begin{cases}
F_x = F \cos \alpha \\
F_y = F \cos \beta \\
F_z = F \cos \gamma
\end{cases}$$
[1.5]

Fig. 1.2

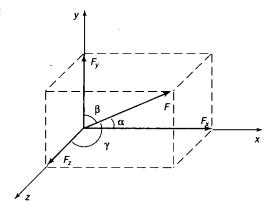
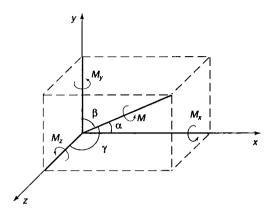


Fig. 1.3



A couple or moment vector about an axis can similarly be resolved into a representative system of component vectors about other axes, as shown in Fig. 1.3 and represented by the following equations:

$$\left. \begin{array}{l}
M_x = M\cos\alpha \\
M_y = M\cos\beta \\
M_z = M\cos\gamma
\end{array} \right\}$$
[1.6]

#### Example 1.1

A set of concurrent forces  $F_i$  are defined by their components  $F_x$ ,  $F_y$ ,  $F_z$  in kN as

Calculate the magnitude and direction of the resultant of this force system.

$$\bar{F}_x = \Sigma F_x = 80 - 40 + 100 - 30 = 110 \text{ kN}$$

$$\bar{F}_y = \Sigma F_y = -20 + 60 - 20 + 10 = 30 \text{ kN}$$

$$\bar{F}_z = \Sigma F_z = -40 - 80 + 30 + 40 = -50 \text{ kN}$$

The magnitude of the resultant force is

$$F_R = \sqrt{(\bar{F}_x^2 + \bar{F}_y^2 + \bar{F}_z^2)} = \sqrt{(110^2 + 30^2 + (-50)^2)} = 124 \text{ kN}$$

to three significant figures. The angles between the resultant force and the axes shown in Fig. 1.2 can now be found using

$$\cos \alpha = \frac{\bar{F}_x}{F_R} = \frac{110}{124} = 0.884$$
, so  $\alpha = 27.9^\circ$ 

$$\cos \beta = \frac{\bar{F}_y}{F_R} = \frac{30}{124} = 0.241$$
, so  $\beta = 76.1^{\circ}$ 

$$\cos \gamma = \frac{\tilde{F}_z}{F_R} = \frac{-50}{124} = -0.402$$
, so  $\gamma = -66.3^\circ$ 

Since all the forces are concurrent, none of them have any moment about the common point through which they all act and there is therefore no resultant moment about that point.

#### Spreadsheet solution

In engineering practice there are few investigations where the problem is as clearly defined and straightforward as that given above. There is usually a degree of uncertainty about the supplied information, some data may be missing and the numerical computations are significantly more complicated. Thus laborious calculations are needed and the sensitivity of the results to variations in the given or assumed information has to be evaluated. This has lead many engineers to develop computer programs in languages such as Fortran, Basic or C to automate the required calculations.

In this book, selected numerical problems will be solved using commercial programming and modelling tools called spreadsheets. Probably the best known of these are Lotus 1–2–3, Borland Quattro and Microsoft Excel. A computer shreadsheet can be considered as a grid of cells which can contain text labels, numbers or formulae which may refer to numbers or numerical results in other cells. If the cell contains a formula, and a number is changed in any cell to which that formula refers, then the formula is automatically recalculated and the updated result is displayed. A range of alternatives can be quickly evaluated by entering different numbers

Fig. 1.4

Α		В	С	D	E
		Fx	Fy	Fz	Magnitude
4	F1	80	-20	-40	
À.	F2	-40	60	-80	
	F3	100	-20	30	
	F4	-30	10	40	AND THE RESERVE OF THE PROPERTY OF THE PROPERT
	FR	+B2+B3+B4+B5	@SUM(C2C5)	@SUM(D2D5)	@SQRT(B7^2+C7^2+D7^2
Direction C	osines				
		alpha	beta	gamma	
		+B7/E7	+C7/E7	+D7/E7	

	A	В	C	Ð	E
1		Fx	Fy	Fz	Magnitude
2	F1	80	-20	-40	
3	F2	-40	60	-80	
4	F3	100	-20	30	
5	F4	-30	10	40	
6					
7	FR	110	30	-50	124.5
8					
9	Direction Cosines				
10		alpha	beta	gamma	
11		0.884	0.241	-0.402	

in a given cell. Modern packages have sophisticated facilities for the graphical display of results, 'What-if' evaluation of a range of alternatives, querying databases and optimization. Although these programs were initially developed for financial modelling, many engineers and students now use these packages as sophisticated programmable calculators.

Figure 1.4(a) shows the data and formulae required to solve this problem. These were entered in a Quattro spreadsheet but they will also work with Lotus 1-2-3 or Microsoft Excel. Figure 1.4(b) shows the resulting display in the spreadsheet. Normally the formulae will not be visible and only the result will be displayed. Throughout this book, the cells in which the user should input data have been highlighted. The other cells contain either formulae or text labels and should not be overwritten.

As can be seen in Fig. 1.4(a), the total force in the x-direction is found in cell B7 by adding up the contents of cells B2, B3, B4 and B5. An even more convenient technique for summing up the y-components is shown in cell C7, where the built-in function @SUM is used to total the contents of cells C2 to C5. Within the spreadsheet program the Edit, Copy and Paste commands can be used to copy the same formula to cell D7, which then sums the z-components. Individual cells or groups of cells can be identified with arrow keys or mouse clicks – it is not necessary to type the cell references. The resultant force is found using the three-dimensional equivalent of eqn. [1.7] in cell E7, using the built-in function @SQRT to calculate the square root and  $^{\wedge}$  to indicate that a number is to be raised to a power.

Once these formulae are entered, any change to any component of the four forces will cause an immediate recalculation of the magnitude of the resultant force and the directions. If only three forces are to be summed, deleting all the components of one force will give the correct answer. If a larger number of forces is to be summed, extra rows can be inserted above row 6 and the SUM formula in the current row 7 adjusted to the new range of cells.

# 1.2 Resultant force and moment

It is sometimes more convenient to replace a system of applied forces by a resultant which of course must have the same effect as those forces. Considering a two-dimensional case as illustrated in Fig. 1.5, then the most general solution is obtained by choosing any point A through which the resultant can act. Then the total force components in the co-ordinate directions are

$$\begin{aligned}
\bar{F}_x &= \Sigma F_x \\
\bar{F}_y &= \Sigma F_y
\end{aligned}$$
[1.7]

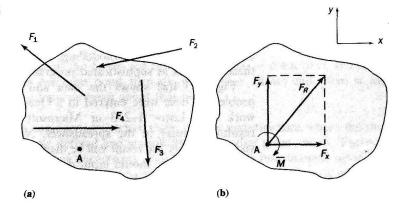
and the resultant force is given by

$$F_R = \sqrt{(\bar{F}_x^2 + \bar{F}_y^2)}$$
 [1.8]

However, this is not sufficient in itself since the moment due to the forces must be represented. This is done by having a couple acting about A such that

$$\bar{M} = \sum M_z \tag{1.9}$$

Fig. 1.5

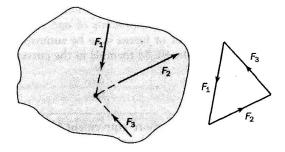


In general, then, any system of forces can be replaced by a resultant force through, and a couple about, any chosen point.

The equivalent solution for a three-dimensional system of forces is similarly a couple and a resultant force whose direction is parallel to the axis of the couple.

One of the most useful constructions in force analysis is termed the triangle of forces. If a body is acted on by three forces then, for equilibrium to exist, these must act through a common point or else there will exist a couple about the point causing the body to rotate. In addition the magnitude and direction of the three force vectors must be such as to form a closed triangle as shown in Fig. 1.6.

Fig. 16



# 1.3 Types of structural and solid-body components

Structures are made up of a series of members of regular shape that have a particular function for load carrying. The shape and function are, through usage, implied in the name attached to the member. The first group is concerned with carrying loads parallel to a longitudinal axis. Examples are shown in Fig. 1.7. A member which prevents two parts of a structure from moving apart is subjected to a pull at each end, or tensile force, and is termed a tie (a). Conversely a slender member which prevents parts of a structure moving towards each other is under compressive force and is termed a strut (b). A vertical member which is perhaps not too slender and supports some of the mass of the structure is called a column (c). A cable (d) is a generally recognized term for a flexible string under tension which connects two bodies. It cannot supply resistance to bending action.

One of the most important of structural members is that which is frequently supported horizontally and carries transverse loading. This is